

$$J(\phi, \psi) = \frac{\partial \phi}{\partial x} \frac{\partial \psi}{\partial y} - \frac{\partial \phi}{\partial y} \frac{\partial \psi}{\partial x} \quad \text{is the}$$

Jacobian determinant

J has the following property

$$J(\phi, \psi) = 0 \Leftrightarrow \phi \text{ and } \psi \text{ are functionally related, i.e., } \phi = F(\psi)$$

In our case, therefore,

$$\Delta \psi = F(\psi) \quad \text{in } \Omega, \quad (13)$$

where $F: \mathbb{R} \rightarrow \mathbb{R}$ must be sufficiently smooth, but is otherwise undetermined

Thus, in 2D steady Euler flows

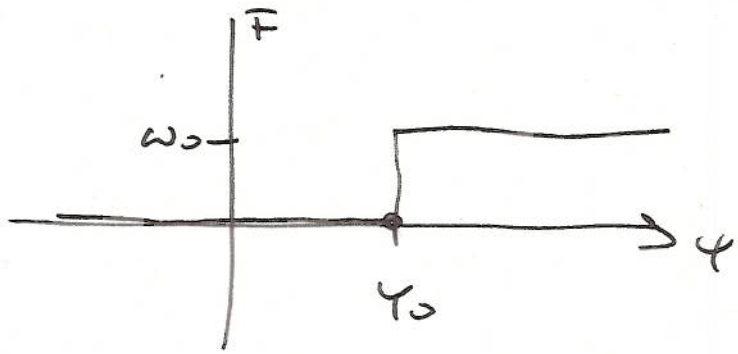
- vorticity is always a function of the streamfunction (constant on streamlines)
- nonuniqueness related to the arbitrariness of F

Examples

Assume

- the flow domain is unbounded $\Omega = \mathbb{R}^2$
- solutions are axisymmetric, $\frac{\partial}{\partial \theta} \equiv 0$, $\psi = \psi(r)$

Let $F(\psi) = \omega_0 [1 - H(\psi_0 - \psi)]$, $H(\cdot)$ - Heaviside (step) function



ω_0, ψ_0 - given

Eq. (13) becomes in the polar coordinates

$$\frac{1}{r} \frac{d}{dr} \left(r \frac{d\psi}{dr} \right) = \omega_0 [1 - H(\psi_0 - \psi(r))]$$

Which is a simple linear ODE and can be solved

Expressing the solution in terms of the azimuthal velocity $u_\theta = -\frac{\partial \psi}{\partial r}$ we have

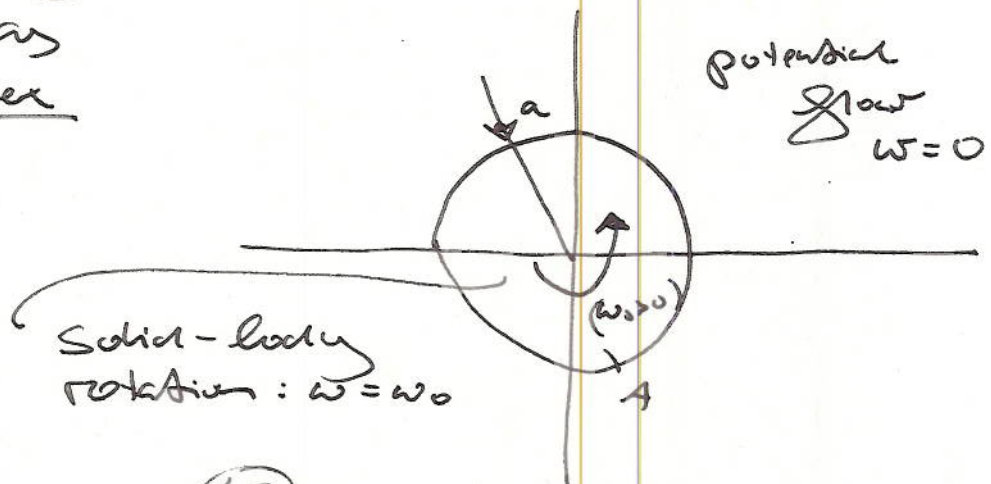
$$u_\theta = \begin{cases} \frac{\omega_0 a^2}{r} & , r \geq a \\ \omega_0 r & , r < a \end{cases}$$

where a is defined such that $\psi(a) = \psi_0$

This is known as the Rankine Vortex

The circulation is

$$\Gamma = \oint \mathbf{S} \cdot d\mathbf{l} = \omega_0 \pi a^2$$



If we fix $\Gamma = \Pi \omega_0 a^2 = \text{const}$ and let $a \rightarrow 0$,
 then $\omega_0 \rightarrow \pm \infty$ and we obtain the point vortex

Ex

Eliminate the assumption $\frac{\partial}{\partial \theta} \neq 0$

Assume $F(\psi) = -k^2 \psi$

Using the ansatz $\psi(r, \theta) = h(r) \sin \theta$, the equation

$$\Delta \psi + k^2 \psi = 0$$

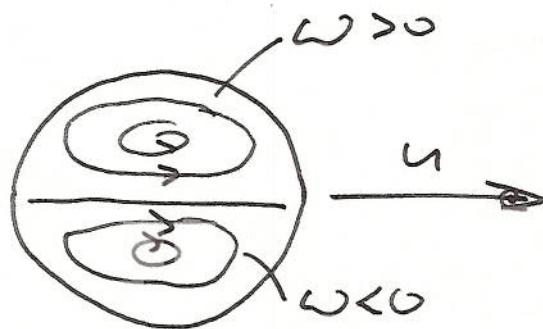
transforms into a Bessel-type ODE for $h(r)$

The solution is therefore $h(r) = C \gamma_1(kr)$, γ_1 - Bessel

$$\left. \begin{aligned} \psi(r, \theta) &= C \sin \theta \gamma_1(kr) \\ \omega(r, \theta) &= C k^2 \sin \theta \gamma_1(kr) \end{aligned} \right\} r \leq a \quad \begin{array}{l} \text{function of} \\ \text{order 1} \end{array} \quad \text{(inside the vortex core)}$$

The constants C and k are fixed to ensure continuity of ψ and ω in the flow domain.

The solution is a propagating vortex dipole

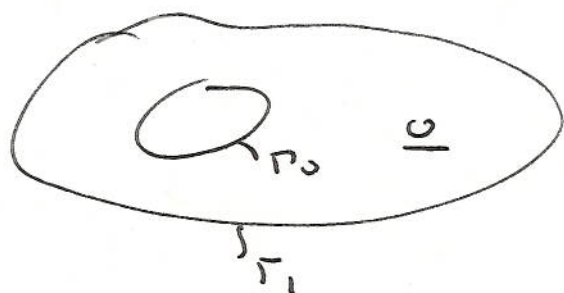


Steady flow
 in the moving
 frame of
 reference

Analogous constructions exist in the 3D axisymmetric
 (steady) case

* More on Potential Flows

Consider a 2D annular region Ω



$$\partial\Omega = \Gamma_0 \cup \Gamma_1$$

$$\bar{u} = \nabla\phi$$

Incompressible flows

$$\left. \begin{aligned} \nabla \cdot \bar{u} &= \Delta\phi = 0 & \text{in } \Omega \\ \frac{\partial\phi}{\partial n} &= u_n^* & \text{on } \partial\Omega \end{aligned} \right\} (*)$$

Functions ϕ satisfying

Laplace's equation are called "harmonic"

$\Rightarrow$ harmonic flows

The theory of such functions is well developed (dating back to pre-computer era), but there are some subtle points.

Are solutions to problem $(*)$ unique?

Suppose $\bar{u}_1 = \nabla\phi_1$ and $\bar{u}_2 = \nabla\phi_2$ are two hypothetical solutions of problem $(*)$.

Denote $\tilde{u} = \bar{u}_2 - \bar{u}_1 = \nabla(\phi_2 - \phi_1) = \nabla\tilde{\phi}$

If sol'n are unique, then $\bar{u}_1 \equiv \bar{u}_2$ and $\tilde{u} \equiv 0$

We ~~also~~ should then have

$$\int_{\Omega} \tilde{u}^2 d\Omega = 0$$

Let's see...

$$\int_0 \tilde{\psi}^2 d\omega = \int_0 (\nabla \tilde{\psi})^2 d\omega = \int_0 [(\nabla \tilde{\psi})^2 + \underbrace{\tilde{\psi} \Delta \tilde{\psi}}_0] d\omega$$

$$= \int_0 \nabla \cdot [\tilde{\psi} \nabla \tilde{\psi}] d\omega = \left| \begin{array}{c} \text{divergence} \\ \text{theorem} \end{array} \right| = \oint_{\partial 0} \tilde{\psi} \frac{\partial \tilde{\psi}}{\partial n} d\sigma = 0 \quad \underline{\text{OK}}$$

But the divergence theorem applies only when $\tilde{\psi}$ is single-valued in 0 . Is it?

What does this mean?

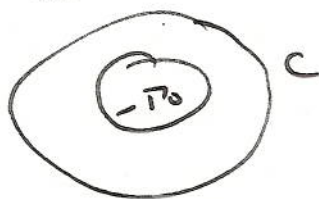
Single-valued $\equiv \oint_C d\psi = 0$ along any contour C in 0

If domain 0 is singly connected, ~~then~~ contour C can be shrunk to a point and $\oint_C d\psi = 0$ is satisfied trivially

Remark

* The exterior of a sphere in 3D is singly connected

* The exterior of a disk in 2D is not singly connected



Conclusion

The potential ϕ defined on a 2D annular domain need not be single-valued. Thus, in such case

$\oint_C (\nabla \phi)^2 ds$ need not vanish and problem (*) may have nonunique solutions.

$$\oint_C d\phi = \oint_C \frac{d\phi}{ds} ds = \oint_C \vec{V} \cdot \vec{t} ds = \Gamma_C$$

To make the solutions unique, we must complement system (*) by prescribing the circulation Γ_C .

If there is circulation, there must be vorticity (Stokes thm), where is it?

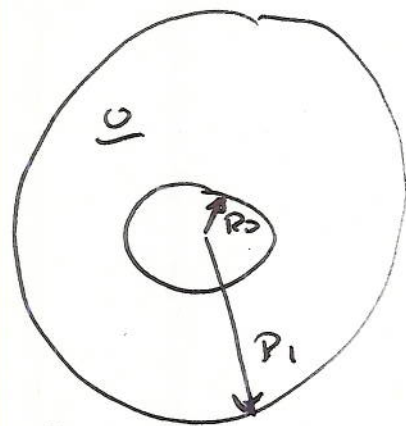
Ex

Consider the flow in an annular domain ~~between~~ between two concentric cylinders with radii R_0 and R_1

$$\vec{u} = \frac{\Gamma}{2\pi} \left(-\frac{y}{r^2} \vec{e}_x + \frac{x}{r^2} \vec{e}_y \right) \quad (\text{Cartesian})$$

$$= \frac{\Gamma}{2\pi} \left(\frac{1}{r} \vec{e}_\theta + 0 \vec{e}_r \right) \quad (\text{polar})$$

$$\vec{u} = \nabla \phi = \frac{1}{r} \frac{\partial \phi}{\partial \theta} \vec{e}_\theta + \frac{\partial \phi}{\partial r} \vec{e}_r$$



} - point vortex with strength (circulation) Γ

Thus:

$$\frac{1}{r} \frac{\partial \phi}{\partial \theta} = \frac{\Gamma}{2\pi r} \Rightarrow \phi(r, \theta) = \frac{\Gamma}{2\pi} \theta + C$$

which is clearly multivalued

Different flows are obtained for different values of Γ (the circulation must be fixed to obtain a unique solution)

* Harmonic Flows in 2D & Complex Variables

There is a deep connection between harmonic flows and the theory of the functions of complex variables

$$\text{Assume } \left. \begin{array}{l} \nabla \cdot \vec{u} = 0 \\ \nabla \times \vec{u} = 0 \end{array} \right\} \text{ almost everywhere in } \underline{\Omega}$$

$$\text{Let } z = x + iy$$

Define the complex potential

$$w: \underline{\Omega} \rightarrow \mathbb{C}$$

$$w(z) = \underbrace{\phi(x, y)}_{\text{potential}} + i \underbrace{\psi(x, y)}_{\text{stream function}}$$

$$i = \sqrt{-1}$$

Note that the velocity components can be expressed as

$$[u, v] = \left[\frac{\partial \phi}{\partial x}, \frac{\partial \phi}{\partial y} \right] = \left[\frac{\partial \psi}{\partial y}, -\frac{\partial \psi}{\partial x} \right]$$

so that $\frac{\partial \phi}{\partial x} = \frac{\partial \psi}{\partial y}$, $\frac{\partial \phi}{\partial y} = -\frac{\partial \psi}{\partial x}$

(Cauchy-Riemann conditions)

Together with some regularity assumptions, these conditions imply that the complex potential is holomorphic (complex differentiable) and analytic.

Thus:

$$(14) \quad \frac{dw}{dz} = V(z) = u(x,y) - i v(x,y)$$

Remark

The Cauchy-Riemann conditions imply that

$$\nabla \phi \cdot \nabla \psi = 0 \quad \text{wherever the gradients are defined}$$

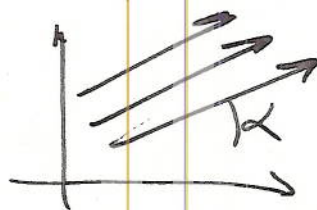
meaning that the streamlines are everywhere orthogonal to the lines of constant potential.

Ex (uniform flow)

$$W(z) = Qz e^{-i\alpha}$$

$$V(z) = \frac{dw}{dz} = Q e^{-i\alpha} = Q (\cos \alpha - i \sin \alpha)$$

$$\Rightarrow \begin{aligned} u &= Q \cos \alpha \\ v &= Q \sin \alpha \end{aligned}$$



Ex (point vortex)

$$W(z) = \frac{\Gamma}{2\pi i} \ln z = -\frac{i\Gamma}{2\pi} \ln z$$

~~velocity components~~ $V(z) = \frac{dw}{dz} = \frac{\Gamma}{2\pi i z} = \frac{-\Gamma}{2\pi} \left(\frac{y}{r^2} + i \frac{x}{r^2} \right)$

known expression for the velocity components of a point vortex

Moreover

$$W(z) = -\frac{i\Gamma}{2\pi} \ln z = -\frac{i\Gamma}{2\pi} (\ln |z| + i(\theta + 2\pi n))$$

take the principal branch ($n=0$)
(multivalued function)

$$= \frac{\Gamma}{2\pi} \theta + i \frac{\Gamma}{2\pi} \ln r$$

$$(r = |z|)$$

potential

stream function

(fundamental sol'n of Laplace eq. in 2D)

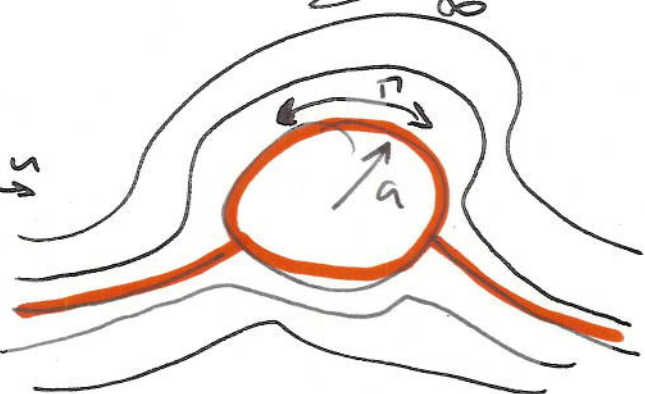
Ex (flow past a rotating cylinder)

$$W(z) = U \left(z + \frac{a^2}{z} \right) - \frac{i\Gamma}{2\pi} \ln z$$

uniform flow at ∞

circle

point vortex



$$\frac{dw}{dz} = u - iv$$

One can check that

$$[u, v] \cdot \vec{n} = 0 \text{ on the boundary}$$